



THALES



AIRBUS



FLUIDIC MANAGEMENT SYSTEM FOR HEMPT-NG

EPIC WORKSHOP FOR ELECTRIC PROPULSION
LONDON 2018

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OPEN



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Project description



Fluidic Management System (FMS) is a **subsystem** of the Highly Efficient Multistage Plasma Thruster - Next Generation (HEMPT-NG) electric propulsion system

Top level objective

- Development, assembly and testing of next generation Fluidic Management System for HEMP-T

Industrial Consortium

- The execution of the subsystem activities are performed by:
 - Thales Alenia Space in Germany &
 - Thales Alenia Space in UK,
 - with AST Space as key supplier.



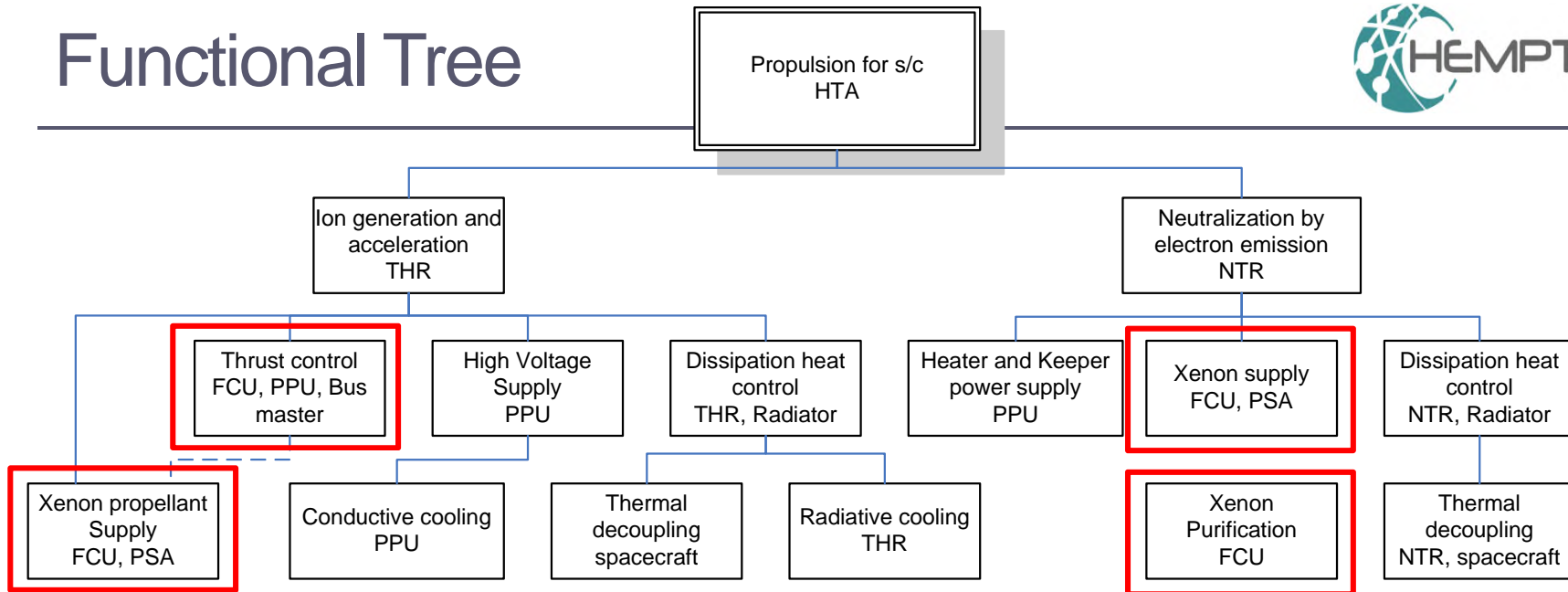
Project major objectives



Major objectives

- Foster innovation: Provide innovative concept, beyond state-of-the-art
 - Usage of **miniature fluidic components**
 - Design having **reduced complexity**
- Create competitive product:
 - **Mass production capability** with reduced manual production steps
 - **Generic products** for several HEMP types
- Increase European non-dependence
 - Development of components with **European origin**

Functional Tree



Needed functions

- Provide Xenon to Thruster and Neutralizer
- Provide means of controlling thrust
- Provide purified gas to neutralizer

Trade-off analysis: Combining PSA and FCU?



PROs	CONs
Only one equipment needed to be tested and qualified	Routing of tubing problematic , since either long high pressure lines or 2 times low pressure tubes lengths
Only one electronics for PSA/FCU control needed	No high flow for cold gas support possible
Reduced complexity	Less flexibility since only one PSA/FCU per thruster

Summary

- In a **radical low cost approach**, the **combination** could make sense. Cold gas is not supported therefore different redundancy concept is needed.
- For **complex architectures** and different amounts of thrusters, the **separation** of PSA and FCU is beneficial

Conclusion

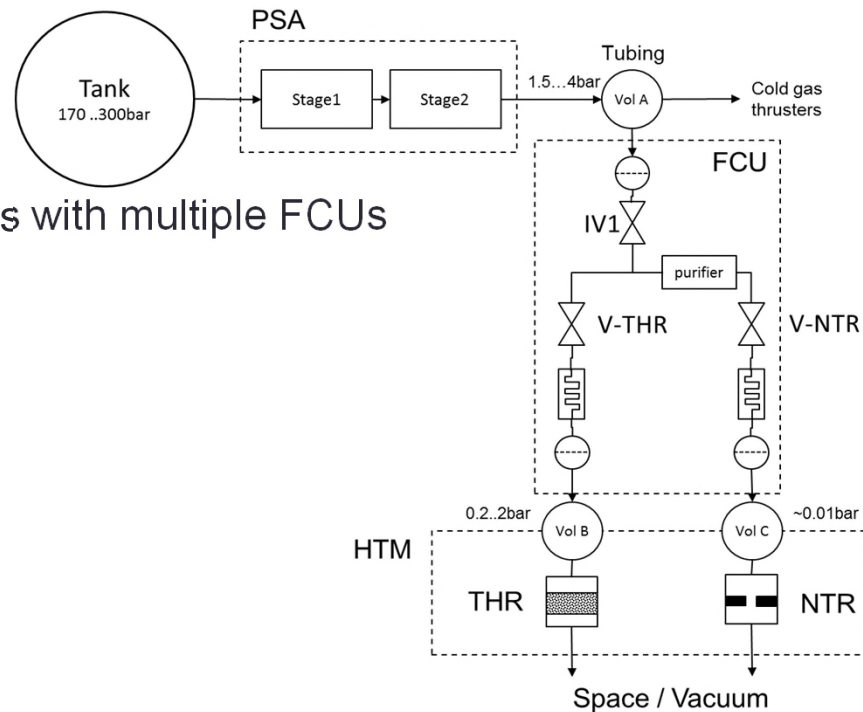
- Since cold gas support is identified as needed in the HEMPT-NG requirements and flexibility (cross strapping) for redundancy is to be used, a **separation of FCU and PSA** is the logical choice.

Design of gas feeding system



Key features

- Designed for **Xenon** or **Krypton**
- Supports **cold gas** thruster
- Supports **cross strapping** of multiple PSAs with multiple FCUs



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Propellant Supply Assembly (PSA)

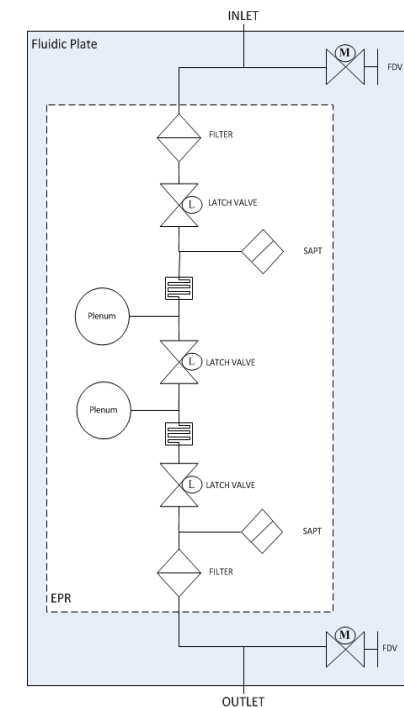


Purpose

- The PSA controls the gas flow from the high pressure side to the low pressure side to adjust the outlet pressure to a set value.

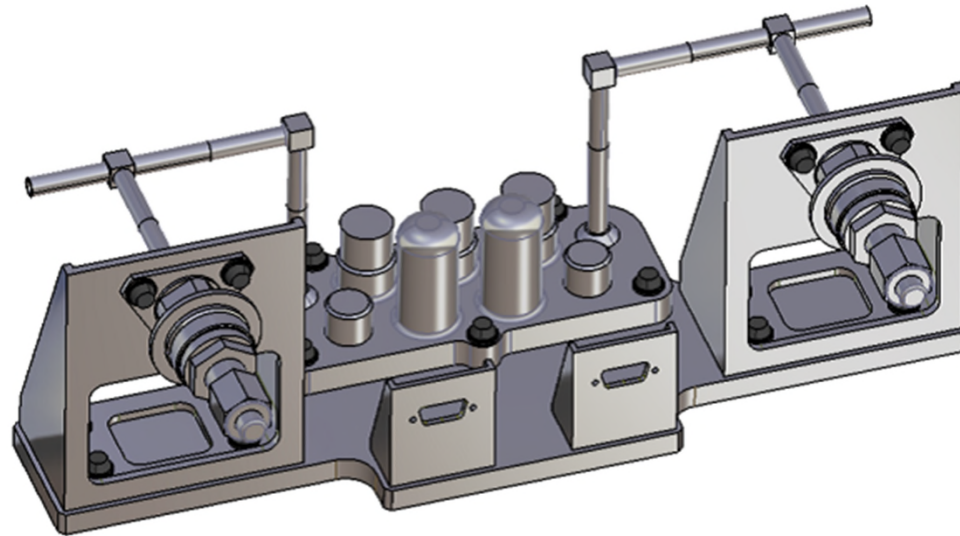
Design

- 1 Electronic Pressure Regulator, made up of:
 - 1 inlet filter,
 - 3 consecutive isolation valves,
 - 2 pressure sensors,
 - 2 plenums,
 - 2 'mazes', integrated into the Flow Path Board, that act as orifices
 - 1 outlet filter.
- 2 Fill and Drain Valves (FDVs),
- 1 set of Titanium tubing,
- 1 Fluidic plate with all equipment integrated on it.



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PSA – CAD model



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PSA – Key performance data



Key features

- **Two stage** pressure regulator (Bang-Bang regulation)
- **Triple barrier** against propellant loss

Parameter	Specified performance
Inlet pressure (MEOP)	200 bar Xe, 350 bar He
Flow range	0...40 mg/s (Electric propulsion) 0...400 mg/s (Cold gas)
Outlet pressure range	0...5 bar
Accuracy / ripple	1.5% (200 mbar / 100 mbar)
Qualification temperatures	20...50 °C (Xenon shall not liquify)
Power consumption	Max. 34.5 W (Typ. <3.6 W EOL)
Internal/External leakage	<10 ⁻⁵ scc/s GHe / <10 ⁻⁸ scc/s GHe
Gas compatibility	Xe, Kr, Ar, He, dry air, N2
Mass	~1.8 kg



Courtesy of AST Space GmbH



Courtesy of Nammo UK

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PSA – Verification Test Sequence for EM



Step #	Verification Test Sequence
1	Initial Inspection and Examination
2	Physical Properties (mass, COI, COG...)
3	Cleanliness and Dryness Verification
4	Proof Pressure
5	Pressure Measurement Calibration Verification
6	Internal & external leakage Tests
7	Evacuation Tests (Venting)
8	Functional Tests
9	Electrical Check-Outs
10	Final examination

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Flow Control Unit (FCU)



Design and architecture

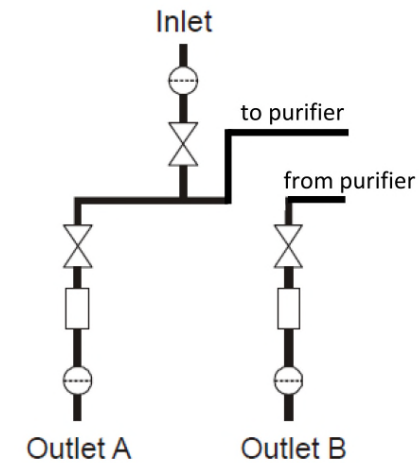
- Includes modified μ FCU with miniaturised components
- Controlled by three valves of one type
- Includes 5 μ m particle filters at inlet & outlet
- 'Mazes', integrated into the Flow Path Board, that act as flow restrictors
- Gas flow to Neutralizer is routed through a gas purifier



Courtesy of
AST Space GmbH

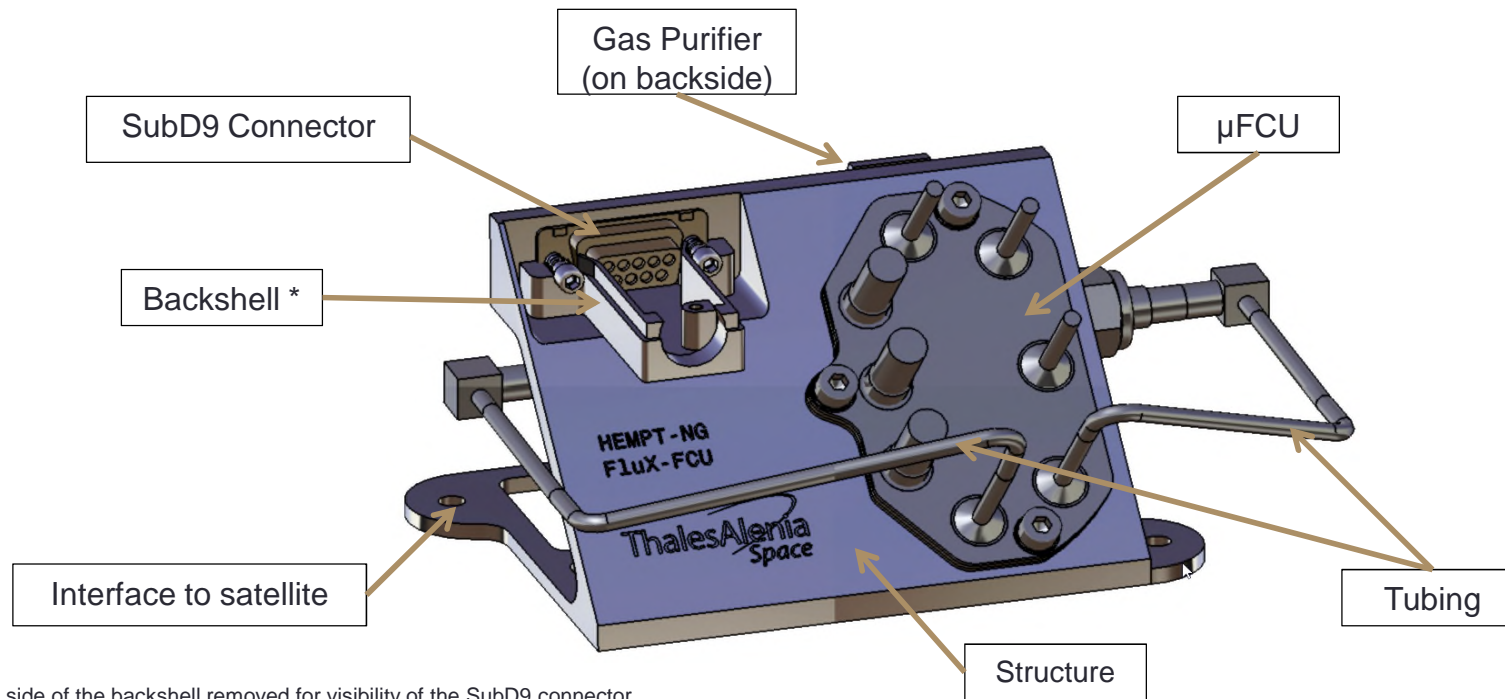


Courtesy of
SEAS Getters



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FCU – CAD model overview



* Top side of the backshell removed for visibility of the SubD9 connector
 Harness not shown in figure

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FCU – Key Performance Table



Parameter	Specified/expected performance
Inlet pressure (MEOP)	2...4 bar (7 bar)
Thruster flow	0,2...2 mg/s (Type LEO)
	2...20 mg/s (Type MEO/GEO)
Neutralizer flow	0,1...0,3 mg/s
Qualification temperatures	-25...90 °C (operating)
Power consumption	Max. 5 W (Typ. <1.5 W)
Internal leakage	<10 ⁻⁵ scc/s GHe
External leakage	<10 ⁻⁸ scc/s GHe over lifetime
Gas compatibility	He, Kr, Xe, N ₂
Mass	~800 g (± 10%)

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FCU – Verification Test Sequence for EM



- **Inspection and Examination**
 - As-built Status & Documentation
- **Physical Characterization Test**
 - Mass & Envelope
 - Interface position and dimensions
- **Electrical Characterization Test**
 - Isolation
 - Coil resistance
 - Power consumption
- **Pressurization Test**
 - Leakage
 - Proof Pressure
- **Functional Performance Test**
 - Flow performance at Inlet pressure extremes
 - Mapping of flow rate as a function of valve duty cycle and frequency adjustment
- **Vibration Test**
 - Resonance search
 - Sine vibration test
 - Random vibration test
 - Mechanical Shock
 - Resonance search
- **Thermal Vacuum Cycling**
 - Non-operational thermal vacuum environment
 - Operational thermal vacuum environment
- **Final Inspection and Examination**
 - As-built Status & Documentation

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FCU – Handling Model



- Check of assembly and handling constraints
- Check of cable lengths and routing



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FCU – Structural Model



- Check of piece parts tolerances
- Check compatibility with mechanical environment



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