



The Strategic Research Clusters on Space Electric Propulsion. A new instrument of the European Commission

EPIC Workshop in London on the 15-17 October 2017

EPIC PSA





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Electrical Propulsion Strategic Research Cluster

What is a Strategic Research Cluster?

Implements multiannual strategic agendas in key research areas Implemented through a system of interconnected grants:

"Programme Support Activity" (PSA): EPIC

- Prepares a roadmap and implementation plan for the whole SRC
- Advices the Commission on definition of calls for operational grants
- Facilitates and supervises the coordination of grants
- Assesses the evolution of operational grants in the SRC context

Several "Operational Grants": OGs

- Address different technological challenges identified in the roadmap.
- Separate projects but with obligation to coordinate/cooperate within the cluster
- The expected results of the individual grants would, when taken together, achieve the overall objective of the SRC.

 EPIC project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 640199

This presentation reflects only the EPIC Consortium's view. The EC/REA are not responsible for any use that may be made of the information it contains.





EPIC Programme Support Activity

















EPIC = Electric Propulsion Innovation and Competitiveness

- EPIC (grant n. 640199) is the PSA project funded as part of the H2020 Space WP 2014+2019; 5+? years duration.
- EP SRC Challenge: to enable major advances in Electric Propulsion (EP) for in-space operations and transportation, in order to contribute to guarantee the leadership through competitiveness and non-dependence of European capabilities in electric propulsion at world level within the 2020-2030 timeframe, always in coherence with the existing and planned developments at national, commercial and ESA level.





EPIC Programme Support Activity

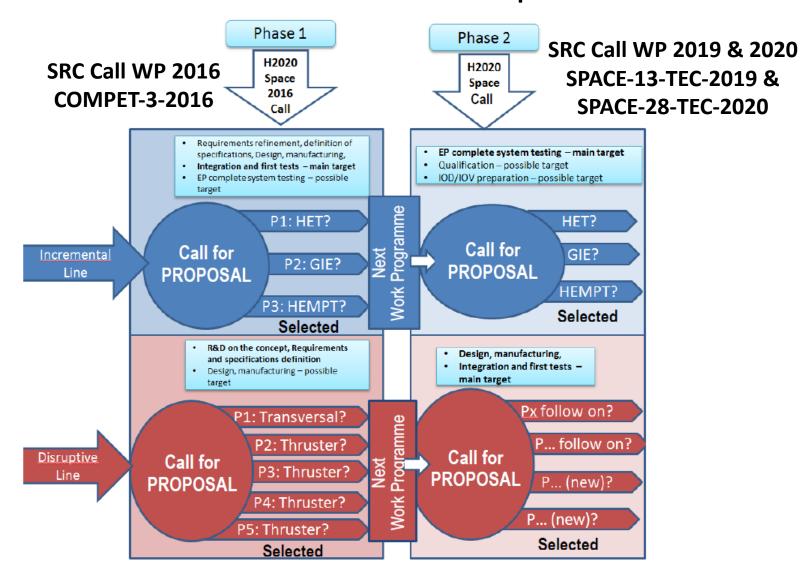
EPIC Main tasks:

- Evaluation on the state of the art and needs of stakeholders
- Definition and refinement of <u>SRC roadmap</u> and master plan for implementation
- Definition of <u>Call topics and related documents</u> for H2020 Work Programmes for funding of SRC Operational Grants
- SRC Risk management
- Definition of the <u>collaboration aspects</u> between SRC grants, including the PSA
- Assessment of the progress and results of the Operational Grants, in the context of the SRC objectives
- Dissemination and education activities





SRC EPIC Roadmap





SRC EPIC Roadmap



Incremental Technologies:

- The Incremental Technologies are the most mature technologies, i.e. the ones with high TRL and possibly with flight heritage, with the physical principal well understood, and with established performances in all of the relevant parameters: thrust (T), specific impulse (Isp), power/thrust ratio (P/T), total impulse, and lifetime.
- They are the Hall Effect Thruster (<u>HET</u>), the Gridded Ion Engines (<u>GIE</u>), and the High Efficiency Multistage Plasma Thrusters (<u>HEMPT</u>).









SRC EPIC Roadmap



Disruptive Technologies:

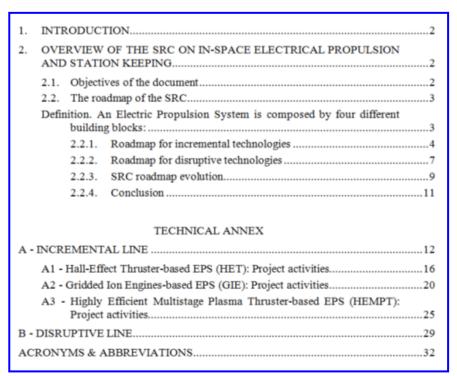
- The Disruptive Technologies, are very promising EP thruster concepts or transversal EP technologies which <u>could disrupt the propulsion</u> <u>sector</u> by providing a radical improvement in performance and/or cost reduction, leading to become the preferred technology for certain applications; or enable new markets.
- Promising EP thrusters are for example: Helicon Plasma Thrusters
 (HPT), Electron Cyclotron Resonance Thrusters (ECRT), Magneto Plasma
 Dynamic Thrusters (MPDT), Pulsed Plasma Thrusters (PPT), Field
 Emission Electric Propulsion thrusters (FEEP), etc.
- <u>Transversal EP technologies</u> are for example radical innovations in Power Processing Units (PPU), magnetic nozzles, alternative propellants, etc.



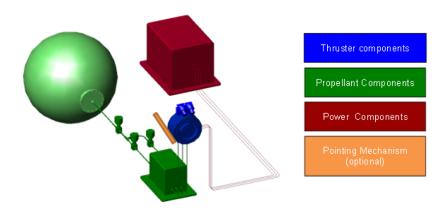


SRC Grants Guidelines & Requirements

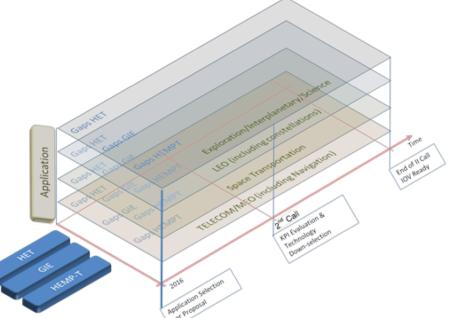
GUIDELINES FOR STRATEGIC RESEARCH CLUSTER ON IN-SPACE ELECTRICAL PROPULSION AND STATION KEEPING HORIZON 2020 SPACE CALL 2016



The Incremental line for the SRC on Electric Propulsion



Electric Propulsion System Main components







SRC Grants Guidelines & Requirements

		Table 1.2							A conflicted on a saledal or	Application activities		
Hall Effect Description and		ET) EPS activities oriented werevolutionary technologies at	* *				Pro o	oposals based on sub-line	Application activities the proposals shall address	Application activities the proposals can choose to address	Applicable Tables	
needed Action	could play a signif Hall Effect Thrust	ystems and applications, and EP pects for use in LEO, due to		Table 2.1			нет	Telecommunications / Navigation LEO Space Transportation / Exploration /		0 1.1 1.2 1.3		
	Projects in this area shall aim at improving EPS p					to			Interplanetary			
Requirements	All HET proposals hereafter.	re cost of the EPS. s shall cover this activity and the	needed Action ma	n and EP is one of the new revolutionary technologies at the moment in satellite					Interplanetary	0, 2.1, 2.2, 2.3 (optional), 2.4 (optional)		
Target TRL				mpetition.	a chemical propulsion as main competitor, and a neitee international				Space Transportation /			
at the end of the COMPET-	5-6		Gr du lau	idded Ion Engin e to their high <i>Is</i> inch costs. Proje	a Engines are one of the best options for this market at the moment high <i>Isp</i> , which allows significant mass savings and allows lower s. Projects in this area shall aim at improving this position in the mid-			НЕМРТ	Telecommunications / Navigation LEO Telecommunications / Navigation	Exploration / Interplanetary Science	0, 3.1, 3.2, 3.3 (optional), 3.4 (optional)	
3-2016				term and at being one step ahead for the future needs of the Telecom market by substantially improving EPS performances and reducing cost Table 3.4							1	
Target TRL at the end of the SRC			Al	GIE proposals shall cover this activity and the requirement reafter. Highly Efficient Multistage Plasma Thruster (HEMPT) EPS to Science applications							_	
(2023/2024)			Target TRL at the			Description and	Science missions can have very specific propulsion requirements. Clear					
if the project were to continue		7-8	end of the COMPET-3-2016 project		5-6	needed Action	examples are the missions requiring micropropulsion with high controllability, for formation flying and high-accuracy orbit control. These missions also require continuous operation for extended periods of time, so they have in addition high Isp and long lifetime requirements.					
• Cycles	TBD by	Due to the eclipses, a large m					This activity is optional for HEMPT proposals.					
	proposers	operation in LEO. Thus, the c the impact that it has on perfc EPS. This number of cycles s lifetime requirement of the pl	(2023/2024) if the project were to		7-8	Requirements Target TRL at the end of the COMPET-3-	This a	This activity is optional for HEMT 1 proposals.				
EPS Power	200-700 W	years) The EPS should demonstrate operated at low to medium po	Dual mode	TBD by proposers	The EPS should be optimized to work in for two different types of functions: EOF	2016 project Target TRL at		6-7				
• P/T • Isp	< 16 W/mN > 1500 s	Low P/T ratio is needed in or little power is available. The EPS efficiency is importa			minimise the time to final orbit; and SK to minimize the propellant used in the in the case of GIE, it is expected that the ef will mainly aim to improve the thrust lev adequate P/T ratio.	the end of the SRC						
		requirement is a trade-off of s	EPS Power	> 5 kW for	The EPS should demonstrate power perf	to continue						
Innovative	Low cost and com	pact PPU		EOR mode	the state of the art, justifying the specific		<1 μN		n low thrust range (<100 μ)	<u> </u>		
and cheaper PPU				> 3 kW for	selected with an analysis of the medium needs.	• Power	< 50 V		Low power levels are expect operation.	ed for micro-propulsion		
EPS Cost	< 200 k€ (indicative)			SK mode	liceus.	Lifetime	> 6 ye		Very long continuous operat	ion	1	
Remarks	Compact, integrate	ed and low mass system shall be	• P/T	~ 21.5 W/mN for EOR mode	The time to orbit is a critical requiremen operators and is fully dependent on the <i>H</i>		> 1000	0 s I f	High <i>Isp</i> is needed, in order for long periods. The higher equirement is a trade-off of	to support continuous operation the <i>Isp</i> the better, but this several performances.		
				~ 30 W/mN		• PPU	needed	d to ensure	high thrust resolution.	e throttability voltage control		
				for SK mode		Remarks	-	arge throttability (1:50) ery low noise				

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SRC 2016 Call Grants CHEOPS

CHEOPS

Consortium for Hall Effect Orbital Propulsion System

- CHEOPS proposes to develop three different Hall Effect Thruster electric propulsion systems: a dual mode EPS for GEO applications, a low power for LEO applications and a >20 kW high thrust EPS for exploration applications.
- Each of these will be developed according to market needs and drivers applying incremental technology changes to existing EPS products.
- Development cover the elements: thruster, cathode, PPU and FMS.
- Objective is to reach at the end of CHEOPS Phase II (2023) the following:
 - TRL7-8 for dual mode and low power
 - TRL6 for high power HET EPS.
- The CHEOPS consortium is led by SAFRAN and is comprised of representatives
 of the biggest European Prime satellite makers, the full EPS supply chain and
 supported by academia.

CHEOPS project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730135







SRC 2016 Call Grants CHEOPS







Partners:

- <u>SAFRAN Aircraft Engines (France)</u>: project coordination, HET Dual Mode System for GEO/NAV and HET system for LEO.
- <u>SITAEL (Italy)</u>: high power HET system for exploration and PPU ir LEO.
- <u>UNIVERSIDAD CARLOS III DE MADRID (Spain)</u>: modelling and transversal activities.
- <u>Thales Alenia Space (Belgium)</u>: GEO dual mode PPU.
- BRADFORD (Netherlands): GEO dual mode FMS and LEO FMS.
- CHALMERS (Sweden): strategies for value creation and cost reduction.
- CNRS (France): modelling, testing and transversal activities.
- OHB (Germany), TAS (France), ADS (France): market analysis, key requirements and specifications elaboration.
- AST (Germany): HET system for exploration FMS
- SME4SPACE (Belgium): dissemination and web site
- DLR (Germany): GEO Dual Mode System MAIT









SME4SPACE









CHEOPS project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730135





SRC 2016 Call Grants GIESEPP

GIESEPP

Gridded Ion Engine Standardised Electric Propulsion Platforms.

- GIESEP proposes to develop, build and test to TRL5 the first European Plug and Play Gridded Ion Engine Standardized to operate ArianeGroup and QinetiQ Space ion engines in the 200-700W (LEO) and 5kW (GEO) domains.
- 5kW electric propulsion system will be designed to allow clustering for 20kW EP Systems for space transportation, exploration and interplanetary missions.
- Dual Mode functionality of the thrusters will be realized, whilst also offering a competitive higher thrust mode.
- Assessments alternative propellants.
- System standardization and the resulting solutions will provide highly cost competitive and innovative EP Systems for current and future satellite markets.
- The activity will also provide the roadmap to higher TRL by 2023-2024, providing a cost competitive EP Systems.

GIESEPP project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730002







SRC 2016 Call Grants GIESEPP

Partners:

- ArianeGroup GmbH (Germany): Coordinator, system engineering, thrusters and testing
- QinetiQ (United Kingdom): System engineering, thrusters and testing
- OHB System (Germany): Satellite system requirements
- <u>CRISA (Spain)</u> and <u>Airbus Defense and Space</u> (<u>Germany</u>): PPUs
- Advanced Space Technologies (Germany): Propellant Control

 Advanced Space Technologies (German
- Mars Space (United Kingdom): Analytical design and test support
- <u>University of Southampton (United Kingdom)</u>: Alternative propellants



CHE

AIRBUS

ariane Group



SRC 2016 Call Grants HEMPT-NG





HEMPT-NG

High Efficiency Multistage Plasma Thruster Next Generation

- HEMPT-NG will develop an integrated EP system based on (Highly Efficient Multistage Plasma Thruster), including the fluidic management system, and the power processing unit.
- HEMPT-NG will offer an ideal EP System for LEO application up to 700 W and for Telecom/Navigation application up 5 kW. The HEMPT technology offers: No discharge channel erosion leading to higher lifetimes of the thruster; Acceleration voltages enabling a high specific Impulse (Isp) leading to a drastic reduction of propellant consumption; Unique large range of thrust offer enormous flexibility; Minimal complexity of concept providing an excellent basis for economic competitiveness.







SRC 2016 Call Grants HEMPT-NG

Partners:

- HEMPT-NG consortium is led by <u>Thales Deutschland GmbH (Germany)</u>,
 Coordinator and responsible for thruster equipment and integrated EP Systems.
- European industrial partners are: <u>Thales Alenia Space (France, Belgium, Germany and UK)</u>, <u>OHB System (Germany)</u>, <u>Airbus Defense and Space (Germany)</u> and <u>Aerospazio Tecnologie (Italy)</u>, who bring their expertise in spacecraft mission studies, equipment development and testing capacities.
- The <u>University of Greifswald (Germany)</u> will provide plasma simulation to support the thrusters developed.













HEMPT-NG project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730020





SRC 2016 Call Grants GANOMIC

GANOMIC

GaN in **O**ne **M**odule Integrated **C**onverter for EP systems

- The consortium plans to build a highly integrated PPU to globally reduce the cost of EP systems.
- Ganomic activities propose focus on the PPU "heart" studying a disruptive power converter beyond the state of the art combining innovative technologies such as Gan digital control, adaptive filtering and embedded packaging.
- The Consortium plans to demonstrate the selected technologies by means of a 7.5 kW power converter to be tested.
- Improvements are expected in cost, mass and volume targeting part list reduction (by 3), converter efficiency (98%) and optimized thermal characteristics (200°C).
- Technical basis for future Direct Drive configurations and to "distributed" configurations where the PPU can be eliminated.





SRC 2016 Call Grants GANOMIC

Partners:

- <u>Safran Electronics & Defense (France)</u> participation to GANOMICS will be overall and technical management of the consortium.
- <u>SITAEL (Italy)</u> participation to GANOMICS will be focused on the integration and testing of the breadboard developed within the project
- <u>Ampère laboratory (France)</u>: The staff included in the project works in the Energy Department which main focus is the integration of power systems
- SAFRAN (Technology Center) (France): virtual prototyping of power assemblies
- <u>Technische Universität Berlin (TUB) (Germany)</u>: research and development in the area of microelectronic packaging and system miniaturization technologies
- <u>UMI-LN2 (France)</u> contributes to provide technical support around GaN Switch integration and PCB embedding & packaging.















SRC 2016 Call Grants HIPERLOC-EP





HIPERLOC-EP

- The HiperLoc-EP project aim to use a novel approach to develop an Electrospray Colloid Electric Propulsion System (ECEPS).
- The project seeks to develop a disruptive electric propulsion technology that provides a high performance EP system a cost that is at least one order of magnitude lower than today. A High Performance Low Cost Electric Propulsion system would enhance the functionality, performance and the value of many micro/nanosatellite missions in the future.
- The objectives include identifying the performance requirements, enhancing the TRL for an ECEPS system, and understanding key processes in order to determine the optimal way to operate an ECEPS.
- The HiperLoc-EP system is anticipated to operate at an efficiency of 50% at an Isp of up to 2500s. The cost target for HiperLoc-EP is to be attractive to constellations of small satellites, CubeSats and nanosatellites.



SRC 2016 Call Grants **HIPERLOC-EP**





Partners:

- Queen Mary Univ. of London (United Kingdom) (Coordinator) provides the leading understanding and expertise in Europe of electrospray processes and systems.
- Systematic (Netherlands) is a IC design house with focus on analog and mixed signal integrated circuit. Has delivered power supply and control circuitry to the Delphi C3 nanosatellite.
- Airbus Defense & Space (United Kingdom) Europe's leading satellite prime contractors and a recognized expert within the field of electric propulsion and as a user of such systems.
- NanoSpace AB (Sweden), has expertise in miniaturized propulsion systems and was among the first to fly a propulsion system onboard a CubeSat in 2015.









HIPERLOC-EP project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730075





SRC 2016 Call Grants MINOTOR





MINOTOR

MagnetIc NOzzle thruster with elecTron cyclOtron Resonance

- MINOTOR's strategic objective is to demonstrate the feasibility of the ECRA
 (Electron Cyclotron Resonance Accelerator) technology as a disruptive gamechanger in electric propulsion, and to prepare roadmaps paving future's way.
- The main objective of the project is to bring the ECRA technology from TRL3 to TRL4/5, in order to demonstrate its potential in a large range of thrust levels.
- ECRA is a cathodeless thruster with magnetic nozzle, allowing thrust vectoring.
 It has a significant advantage in terms of global system cost and reliability compared to mature technologies. It is also scalable and can potentially be considered for all electric propulsion applications, from microsatellites to space tugs.

MINOTOR project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730028











Partners:

- ONERA (France). Coordinator, and in charge of most experimental investigations of the thruster configuration.
- <u>University Carlos III de Madrid (Spain)</u> will develop the codes and implement the numerical modeling of the thruster.
- <u>Thales Microelectronics (France)</u> will demonstrate a high efficiency microwave generator technology.
- <u>Universitate Giessen (Germany)</u> will conduct the higher power tests (1 kW) and the erosion test on the 200 W prototype.
- <u>Thales Alenia Space Belgium SA (Belgium)</u> will investigate the impact of the ECRA technology on the PPU architecture and cost.
- <u>Safran Aircraft Engines (France)</u> will provide expertise in electric propulsion thruster production and performance.
- L-up (France) will help on the project management.















MINOTOR project has received funding from the European Union's Horizon 2020 research and innovation programme under grant agreement No 730028



SRC Next Steps



- Assessment of the progress and results of the ongoing Operational Grants.
- <u>Update</u> the SRC <u>EPIC roadmap</u> and master plan.
- Preparation of the second phase of the SRC EPIC roadmap: SRC 2019 Call topic (SPACE-13-TEC-2019) and SRC 2020 Call topic (SPACE-28-TEC-2020).
- Dissemination and educational activities.
- <u>EPIC Workshops</u>: London (United Kingdom) on 2018, and Noordwijk (Netherlands) on 2019.
- <u>EPIC Lecture Series</u> in concurrence with the next EPIC Workshops.