

The CO2DGM for CO2 – Breathing Thrusters

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Abstract (1/2)

♣ CO2DGM is a Detailed Global Model of plasma generated from a gas constituted initially by CO2

The effective plasma composition depends on various imposed conditions, e.g. form factor / pressure / absorbed power

CO2DGM is used here to support various types of Electric Thruster (ET) technology, allowing for:

- Thruster characterization
- Functioning description and optimization
- Optical Emission Spectroscopy (OES) diagnostics

With **CO2DGM** the plasma components constitution of CO2 fed **ETs** and their spectra are obtained. The ability of **CO2DGM** to characterize and to diagnose electric thrusters is illustrated.



Abstract (2/2)

- ** LMO / HMO In Situ Resources Utilization (ISRU) ETs fed by Mars Atmosphere components **
- Plasma Component Composition (PCC) results
- Functioning Diagrams (FD) results
- ♣ Theoretical intensities of O I & II atomic lines (OES)

Characterization / diagnostics of CO2 fed ETs in a large domain of plasma conditions are addressed

One of the **CO2DGM** assets is the calculation of the spectral lines intensities.

For **OES** diagnostics calculated intensities of the spectral lines must be compared with the **experimental** ones from the ET plasma.



Summary of Presentation

- 1. INTRODUCTION including References
- 2. DENSITIES OF SPECIES based on PCC and conjugate diagrams
- 3. IONIZATION PERCENTAGE based on FD diagrams
- 4. OES PLASMA DIAGNOSTICS based on THEORETICAL OXYGEN I & II SPECTRA
- 5. CONCLUSIONS



1. Introduction (1/2)

- Our CO2DGM constitutes a tool in support of CO2 fed electric thrusters of various types
- ♣ ISRU technology is specifically addressed by adapting accordingly the input parameters
- ♣ We approach OES diagnostics of CO2 fed devices

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1. Introduction (2/2)

- 1. Katsonis, K. & Berenguer, Ch. (2014). Characterization of Low Pressure CO₂ Plasma in Space. ESA 6th RHTG Works., St. Andrews, UK
- 2. Berenguer, Ch. & Katsonis, K. (2014). Global Modeling of CO2 Discharges with Aerospace Applications. Advances in Aerospace Engineering. Vol. 2014, Article ID 847097
- 3. Katsonis, K., Berenguer, Ch., Gonzalez del Amo, J. & Stavrinidis, C. (2016). CO2 / N2 Breathing Electric Thrusters for LMOs, Space Prop. Conf., Paper ID SP2016 **3124972** Rome, Italy
- 4. Berenguer, Ch. & Katsonis, K. (2016). A Detailed Global Model for Characterization of CO₂ Fed Electric Thrusters, Imperial Journal of Interdisciplinary Research 2, 1708
- 5. Katsonis, K. & Berenguer, Ch. (2016). Study of CO2 Plasma of Interest to Space Applications Based on a **Detailed Global Model**, 7th ESA RHTG Workshop, November 2016, Stuttgart, Germany
- 6. Katsonis, K. & Berenguer, Ch. (2018). Characterization and Optical Diagnostics of CO2 Fed Electric Thrusters by Using a Detailed Global Model, 6th Space Prop. Conf., Paper ID SP2018 00237, Seville, Spain; Summary presentation in Poster.
- 7. Katsonis, K. & Berenguer, Ch., ISRU based electric propulsion, ResearchGate Project

Fig. 1.

Pressure
dependent
PCC for
= 500 W

CO2 feeding of 20 sccm

Only **C**containing
components
shown

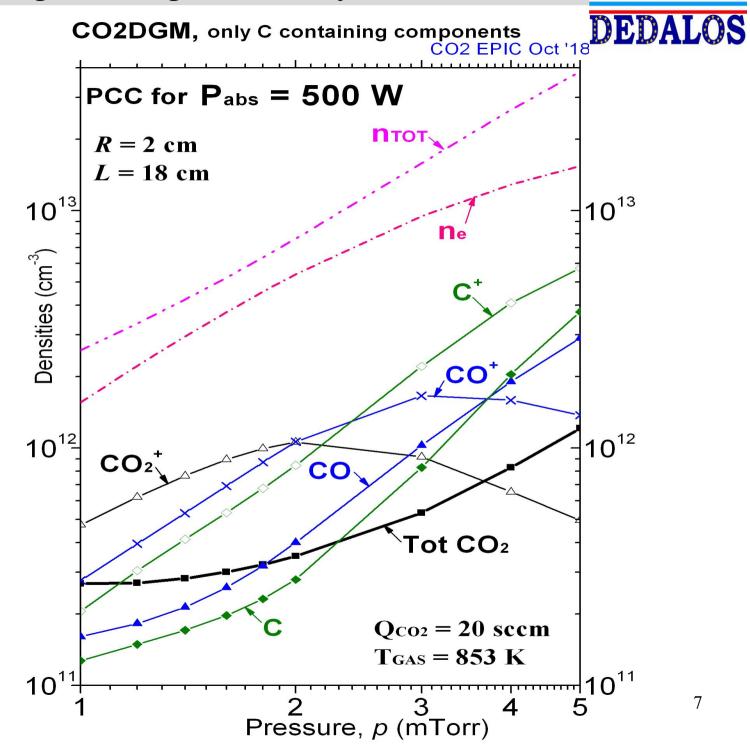


Fig. 2.

Pressure
dependent
PCC for
= 200 W

CO2 feeding of 20 sccm

Only **C**containing
components
shown

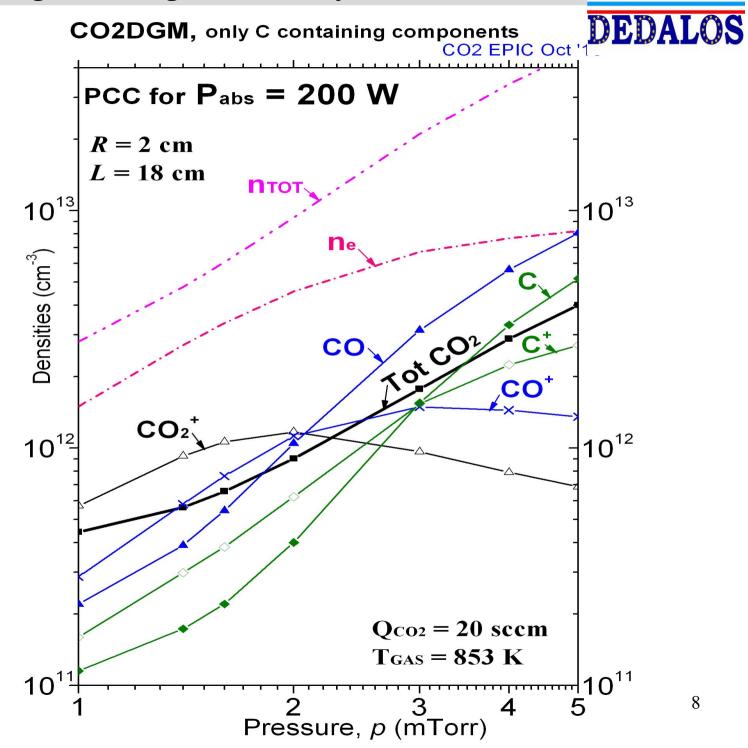


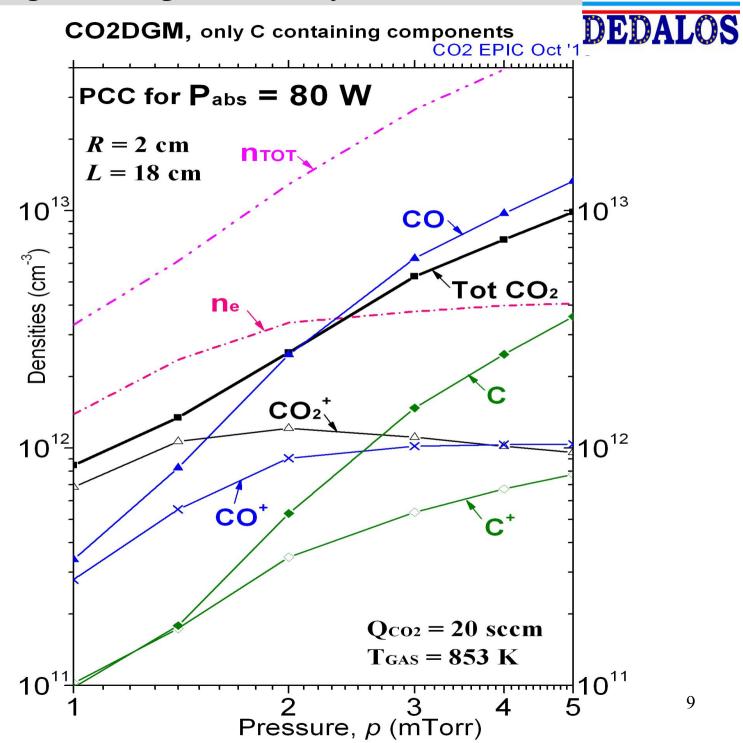
Fig. 3.

Pressure
dependent
PCC for
Pabs

P 80 W

CO2 feeding of 20 sccm

Only **C**containing
components
shown



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2. DENSITY OF SPECIES, **PCC**

Fig. 4. **Pressure** dependent PCC for $P_{abs} = 500 W$

CO₂ feeding of 20 sccm

Only components without C shown

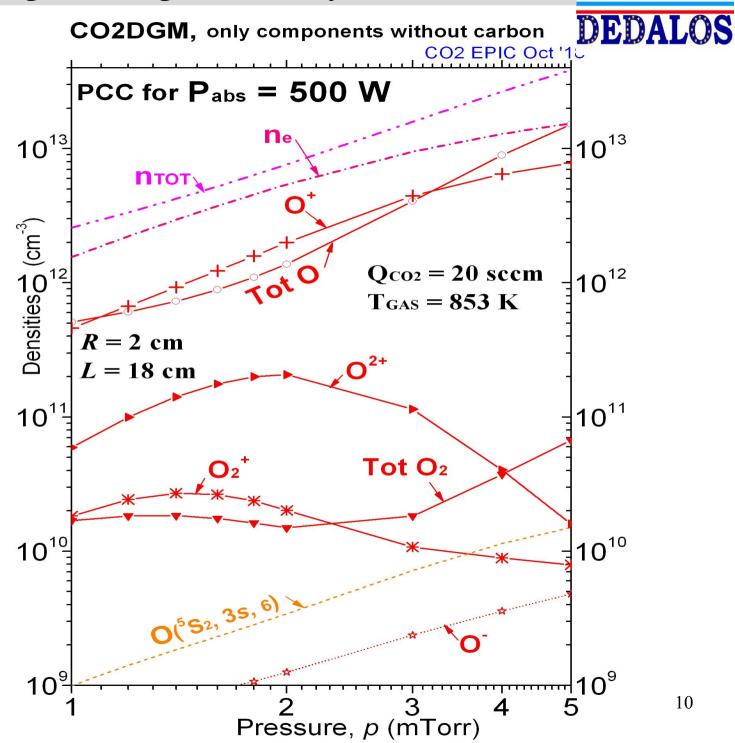
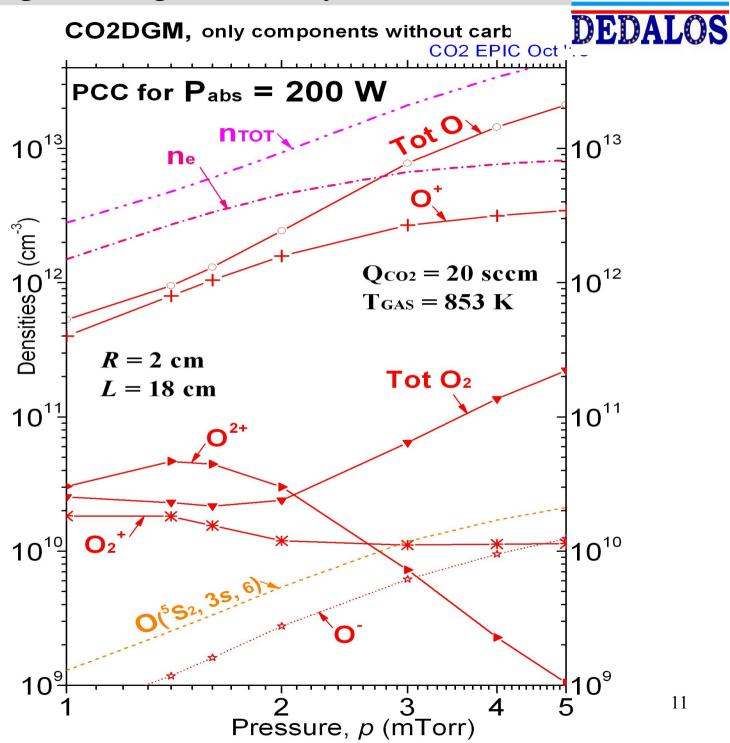


Fig. 5.

Pressure
dependent
PCC for
Pabs

CO2 feeding of 20 sccm

Only components without **C** shown



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2. DENSITY OF SPECIES, **PCC**

Fig. 6. **Pressure** dependent PCC for $P_{abs} = 80 W$

CO₂ feeding of 20 sccm

Only components without C shown

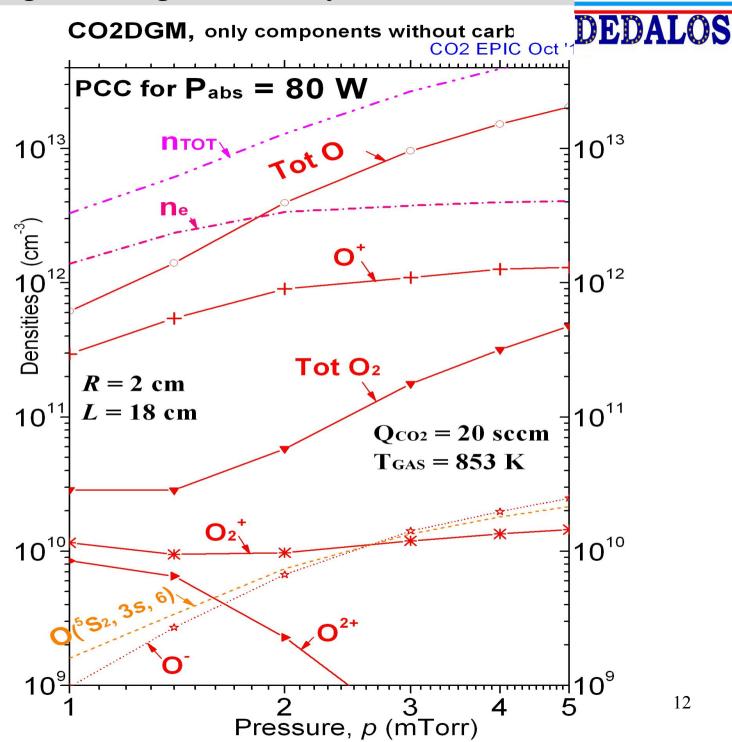




Fig. 7. Conjugate of Figs. 1 & 4

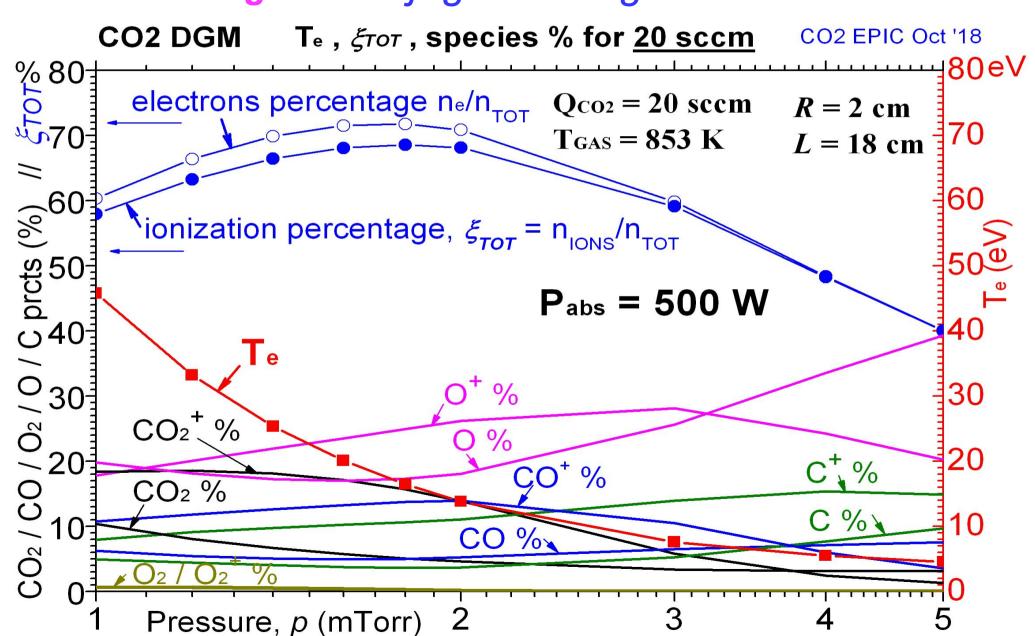




Fig. 8. Conjugate of Figs. 2 & 5

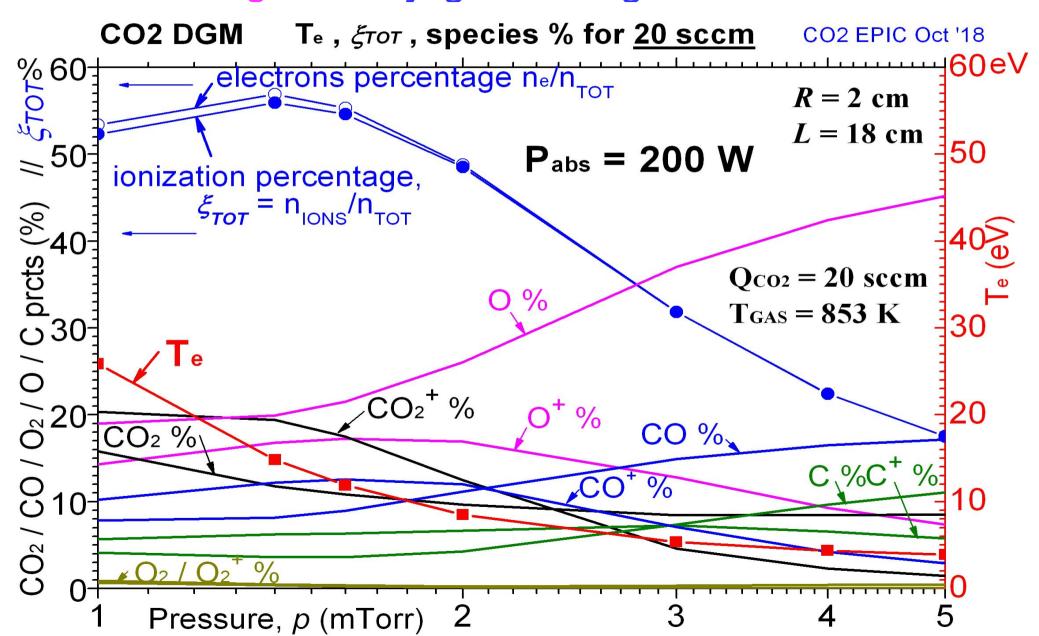
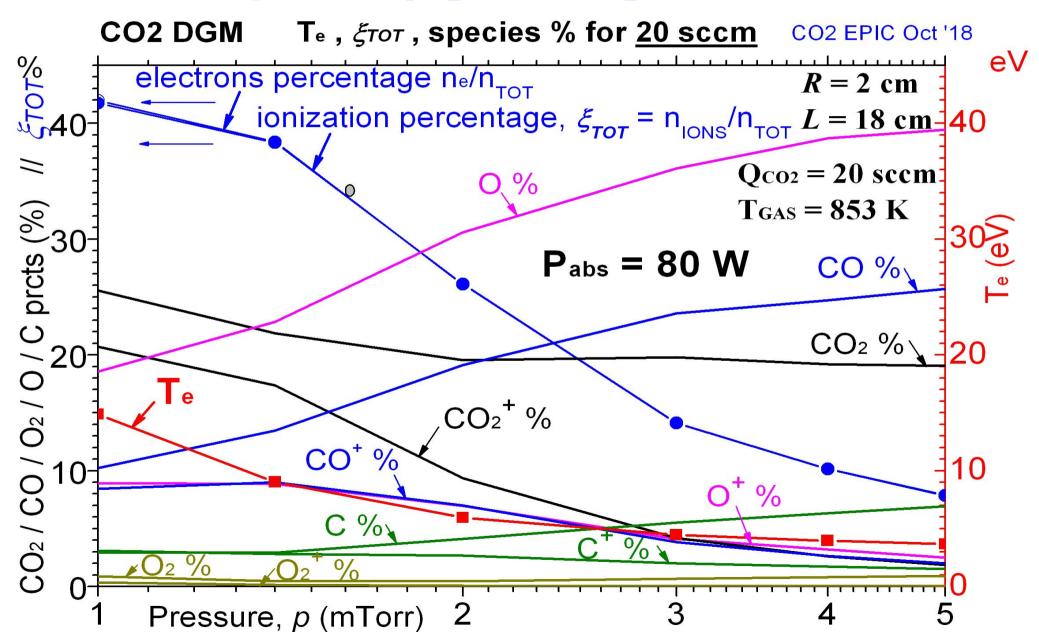




Fig. 9. Conjugate of Figs. 3 & 6



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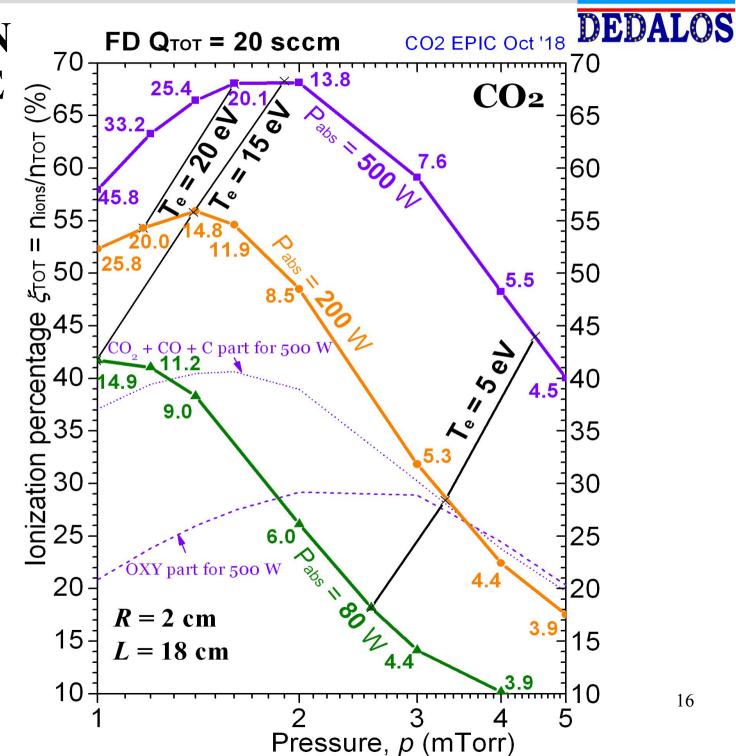
3. IONIZATION **PERCENTAGE**

Fig. 10. FD for p values from 1 mTorr up to 5 mTorr

T_e of 5 - 20 eV

P_{abs} of 80 W - 500 W

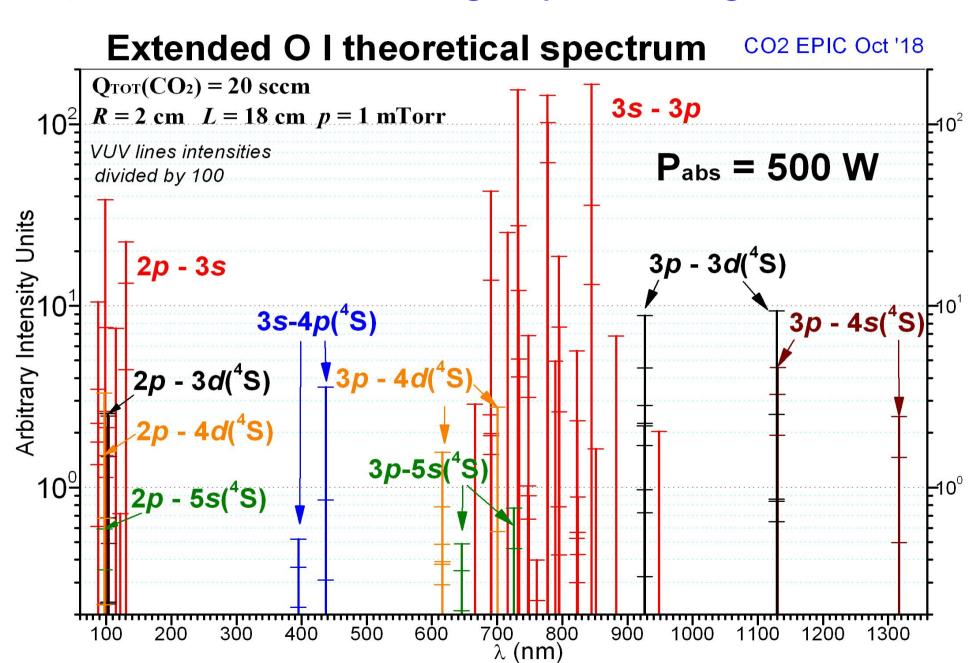
20 sccm **CO**2 feed



4. OXYGEN I SPECTRUM (Theoretical)



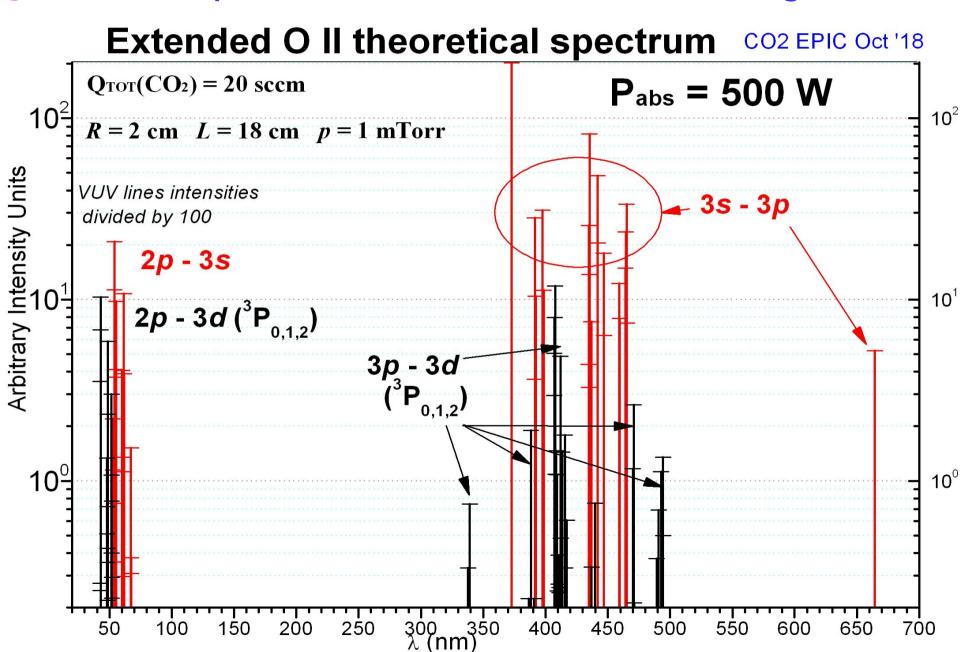
Fig. 11. O I lines in a large spectral region



4. OXYGEN II SPECTRUM (Theoretical) DEDALOS



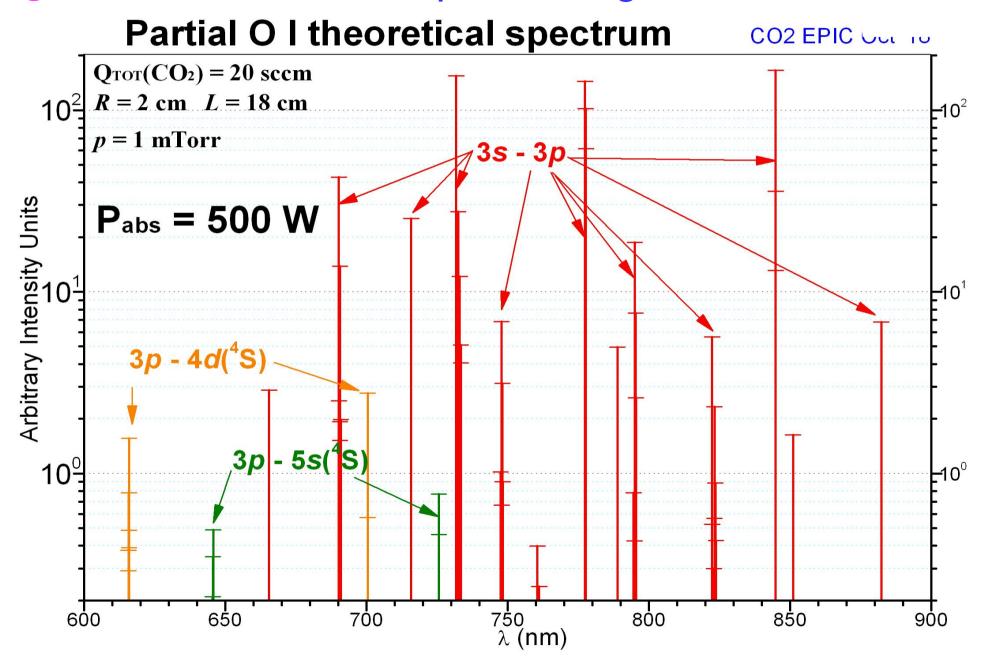
Fig. 12. O II spectrum, VUV, UV & visible regions



4. OXYGEN I SPECTRUM (Theoretical)



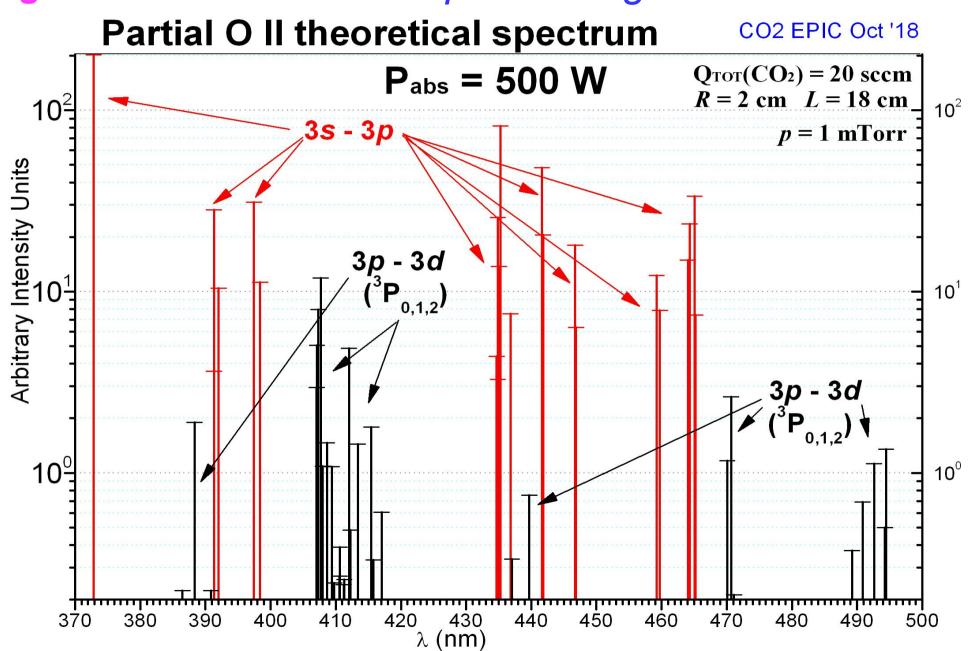
Fig. 13. O I lines in the spectral region 600 - 900 nm



4. OXYGEN II SPECTRUM (Theoretical) DEDALOS



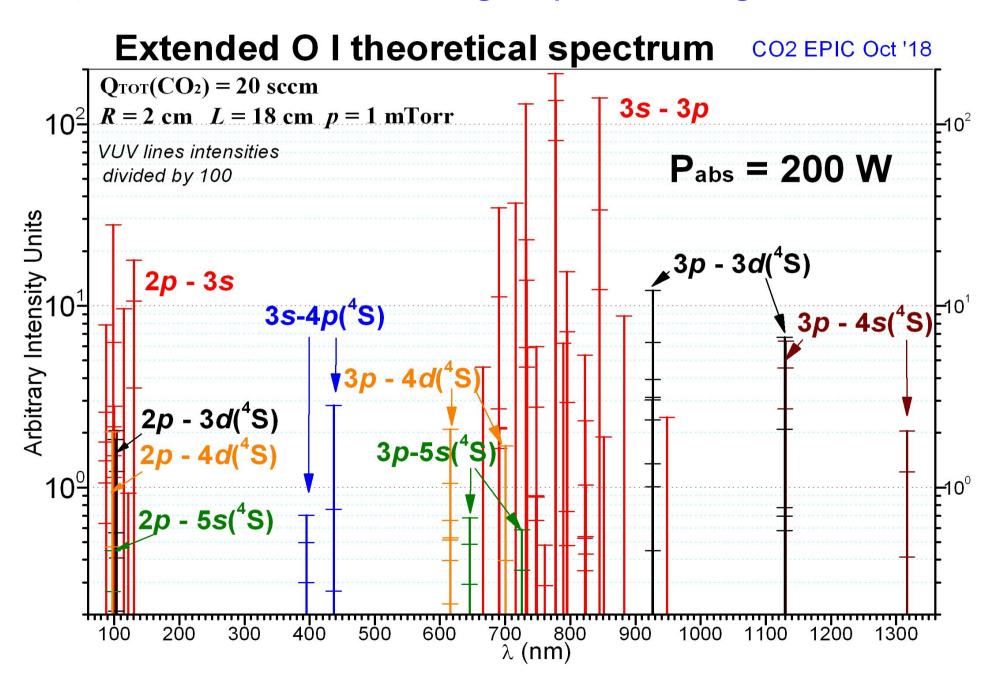
Fig. 14. O II lines in the spectral region 370 - 500 nm



4. OXYGEN I SPECTRUM (Theoretical)



Fig. 15. O I lines in a large spectral region

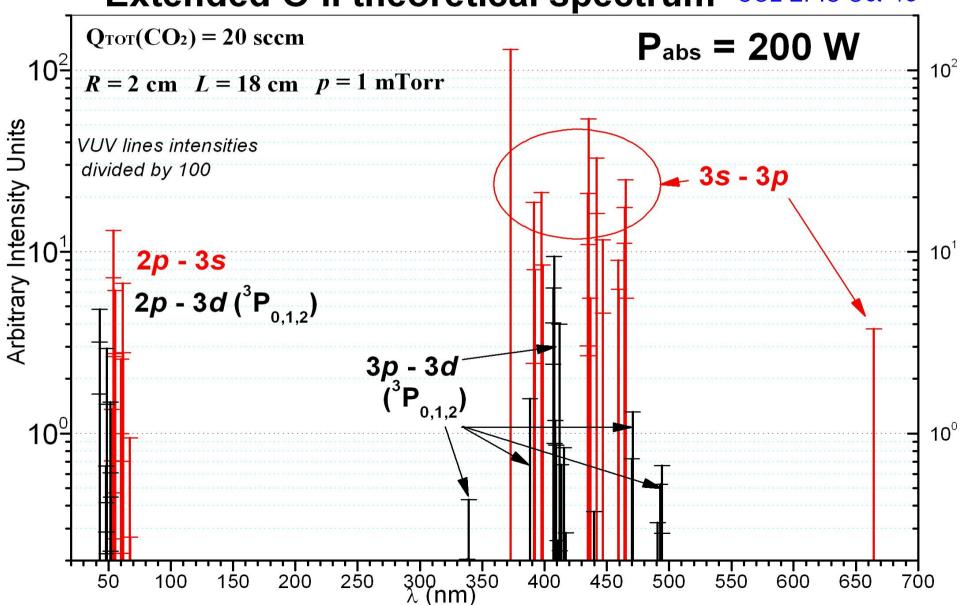


4. OXYGEN II SPECTRUM (Theoretical) DEDALOS



Fig. 16. O II spectrum, VUV, UV & visible regions

Extended O II theoretical spectrum CO2 EPIC Oct '18





5. CONCLUSIONS

We conclude that **CO2DGM** contributes efficiently to theoretical characterization and diagnostics of ETs fed by **CO2** by using **PCC** and **FD** diagrams and theoretical spectra.

Various absorbed power values have been addressed for 20 sccm of CO_2 flow rate and a form factor of R = 2 cm, L = 18 cm.

Work presented here extends in a lower Pabs domain the work presented recently in Seville [6].

Thank you for your attention